Effective Properties of Smart Elastic Laminates and the Screening Phenomenon

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Abstract

Non-stationary phenomena arising in structures can be effectively controlled via smart (intelligent) materials. The corresponding mathematical problems contain controls in the coefficients of hyperbolic equations, these controls depending both on position and time. When appropriately activated, such controls may provide 100% screening of some extended parts of the structure from the invasion of dynamic disturbances. For bars exposed to longitudinal vibrations, the material pattern producing such a screening effect can be the combination of rank-one laminar composites in space-time. Such a composite is described as an array of alternating segments with different pairs (ρ_1, k_1) and (ρ_2, k_2) of density and stiffness of smart material, this whole array traveling along the bar with a suitable constant speed V.

1. Introduction

The appearance of smart materials has made it realistic to claim for the structures with properties that may vary both in space and time, thus making these structures highly responsive to the non-stationary environment. We shall place emphasis on the design of material patterns able to block the propagation of undesirable disturbances caused by impulsive loads, impacts, etc., into some extended parts of the structure vulnerable to a material damage. Such screening may be achieved through the deployment of special assemblages of smart materials, e.g. smart laminar composites, activated at appropriate location at appropriate time.

Mathematically, the problem reduces to that of the control of coefficients in hyperbolic equations both in space and time. This control is assumed to be materialized through the sensors and actuators distributed throughout a smart structure.

2. Basic Conceptual Features of Smart Material Design

The above design concept can be efficiently implemented through the use of smart materials. A smart material is assumed to have the capacity to change its properties both in space and time to suit the demands placed upon it by the environment.

The smart material is constructed using a distributed set of sensors and actuators embedded within the body of a structure. The optimal design of structures based on use of smart materials will be called *active optimal design*. Some recent advances in a smart material design are summarized in Anderson & Tsou (1992).

Examples of smart materials are:

- (i) shape memory alloys;
- (ii) composites featuring electrorheological fluids that possess the ability to alter their viscous properties under the presence of an electric field;
- (iii) components made by embedding piezoceramic actuators and sensors into a matrix.

With ordinary (non-smart) materials, the design is necessarily passive and restrained. Indeed, if the goal is to create a material able to feature a desired response to a variety of external conditions which may be competitive or even contradictory, then structural resources will split to meet the set of requirements rather than to become entirely focused on the immediate environmental demand.

The concept of smart (active) material design is entirely different. Here, the system responds at each time to the immediate demand required by the environment. The effectiveness and flexibility of this design strategy is incomparatively greater than those of the passive approach. Conceptually speaking, the smart materials allow the designer to restrict himself with single purpose design instead of multi-purpose design. The structure will resemble a living tissue insofar as it is able to mobilize its properties to meet the immediate environmental demands. Another benefit associated with smartness is the ability to manipulate several physical properties independently.

The concept of active design is implemented through the assumed dependence of controls both on space and time. In the context of structure mechanics, this concept has so far been addressed with respect to external loads viewed as controls.

Probably the earliest work on this subject has been initiated by Butkovsky et al (see Butkovsky et al (1974), Butkovsky (1982), summarized in Butkovsky & Pustilnikov (1980)). These authors introduced what they called "movable controls" to effectively suppress vibrations. Eventually, a similar concept has been exercised in McLaughlin & Slemrod (1986). A moving ("scanning") control is the one concentrated within a certain spatial domain, this domain moving through the space as a solid body. This idea works very well when applied to external sources (loads), it must work even better when implemented toward material or structural parameters.

The elastic moduli (stiffnesses) of structures are known to be very effective controllers. By now, there is plenty of information concerning optimal layout of materials differing in their elastic stiffnesses. This information is related either to statics, or to problems of free vibrations or loss of static stability, in all cases are materials assumed to be passive, i.e. their stiffnesses never being time dependent.

The concept of general space and time dependent stiffnesses is intrinsic in the idea of smart materials, and it promises substantially new possibilities of controlling the systems' behavior. Specifically, with such stiffnesses combined with space and time dependent material density, it becomes possible to selectively screen large domains in space-time from unwanted disturbances which could be initiated either by external influences (i.e., impacts), or by permanently applied loads. The following examples will illustrate this idea mathematically.

3. Control of the First Order Evolution Equation

Consider a controlled system governed by the first order equation (Lurie (1971))

$$z_t + uz_x = 0, \quad 0 \le t < \infty, \quad -\infty < x < \infty, \tag{1}$$

with u = u(x, t) treated as control.

Complemented by the initial condition

$$z(0,x) = z_0(x), \quad -\infty < x < \infty, \tag{2}$$

this equation describes signals propagating with the phase velocity u = u(x,t).

Suppose that u = u(x,t) takes the values $u_1, u_2(u_1 < u_2)$ as shown on Fig. 1:

$$u = \begin{cases} u_1 & \text{in parallelogram } abcd, \\ u_2 & \text{otherwise.} \end{cases}$$
 (3)

In other words, assume that the line segment ab describing the zone within which the phase velocity takes the value u_1 , is itself traveling as a solid body with uniform velocity U (see Fig. 1). If this velocity (represented as a slope of the parallelogram abcd) satisfies $u_2 > U > u_1$, then the disturbances originating on the initial manifold concentrate on the leeward side ad of the parallelogram. This concentration occurs due to the fact that the disturbances initiated to the left of the segment ad overtake this segment in their subsequent travel. At the same time, the disturbances within the segment lag behind since the segment moves faster. Mathematically, this is expressed as the collision of characteristics arriving at ad from both sides. Because the array $(x_0, x_0 + h)$ of oncoming disturbances has been redirected, no disturbances will enter the quiescent region ("shadow zone") d'dfbb' shown on the figure. This region is thus completely screened from the encroachment of a disturbance z(x,t) which is identically zero within it.

4. Control of the Second Order Hyperbolic Equation

The longitudinal wave propagation along elastic bars is governed by the second order hyperbolic equation of the type

$$(\rho z_t)_t - (kz_x)_x = 0; (4)$$

this equation provides complexities compared with its lst order counterpart (1). The main reason is because Eq. (4) introduces two phase velocities: $a = \sqrt{k/\rho}$ and

 $-a=-\sqrt{k/\rho}$, i.e. the waves traveling in opposite directions. Desiring to suppress both waves in a sense indicated above we should take actions aimed to create waves travelling both in the same direction.

Such waves may be arranged through the formation of composites with parameters that are variable both in space and time.

More specifically, we will assume that the material parameters ρ , k address the following properties:

- (i) they are both space and time dependent;
- (ii)at each point (x,t) the pair (ρ,k) may take either the values (ρ_1,k_1) or the values (ρ_2,k_2) ;
- (iii) these admissible values are taken within alternating layers having the slope dx/dt = V so chosen as to ensure regular transition of continuous disturbance z(x,t) across the interface from one layer to another. In other words, both kinematic and dynamic compatibility conditions will be observed across the interface.

Condition (iii) will be satisfied if we postulate the following relationship between the characteristic slopes $a_i = \sqrt{k_i/\rho_i}$, i = 1, 2, and V (we assume that $a_2 \geq a_1$)

$$\frac{V^2 - a_1^2}{V^2 - a_2^2} \ge 0. {(5)}$$

The procedure that will eventually produce the screening effect applied to Eq. (4) will include two steps. First, we use the properties (i)-(iii) and choose the value of V and the volume fraction m_1 of material (ρ_1, k_1) in a composite to construct a rank 1 laminate with effective phase velocities $a'_1, a'_2(a'_2 \geq a'_1)$ possessing the same sign, say positive, i.e. having disturbances traveling both to the right. We will call such a composite r-composite (Fig. 2a). Applying -V instead of V in the above construction, we create the ℓ -composite, with effective phase velocities $-a'_1, -a'_2$ being both negative and related to waves traveling to the left (Fig. 2b).

At the second step we apply r - and ℓ - composites to construct the material that implements the screening property. To this end assume that the (x,t)-plane is occupied by the r-composite everywhere but within the rectangle mnpq where the ℓ -composite is applied (Fig. 3). The characteristics of both families will then collide at the leeward face mn of the rectangle, and the discontinuity will develop along it. At the same time, within the domain rqsnu the value of z will be identically zero because the initial disturbance z(0,x) will never reach it since it will be averted from it on a halfway due to the appropriate turning of characteristic directions.

This screening effect will remain in place if we allow for the domain mnpq to be a parallelogram with the slope U of faces mn and pq satisfying the condition

$$\mid U \mid \leq \min(a_1', a_2').$$

As before, the screening effect has become possible due to the smartness of participating materials, specifically because of the time dependence of their stiffness and density. Both characteristics may be made space and time dependent in a variety of ways, e.g. through the electrorheological fluid pumped in and sucked out along the bar at appropriate locations at appropriate time. The time dependence of density ρ is intrinsic in the construction specified above: in the absence of such dependence this construction fails to work because the rank 1 composites mentioned above will then possess effective phase velocities of opposite signs. It is still unknown whether the screening effect will ever be possible lest the density be time dependent.

5. Constructing the r-and ℓ -Composites

We make use of conditions (i)-(iii) and Ineq. (5) to apply a standard procedure (Bakhvalov & Panasenko (1984), Lurie (1993)) of calculating the effective phase velocities of a composite assembled from the above mentioned layers distributed in space-time, i.e. on the (x,t)-plane.

Eq. (4) is replaced by the system

$$\rho z_t^1 = z_x^2, \quad k z_x^1 = z_t^2 \tag{6}$$

with $z^1 = z$, and z^2 denoting an auxiliary dependent variable. We will require that both z^1 and z^2 be continuous, thus ensuring the conditions of kinematic and dynamic compatibility, respectively, to hold. Formally, this means particularly that Ineq. (5) should be satisfied.

Let us introduce coordinates τ, n , measured, respectively, as distances along and across the layers on the (x,t)-plane; the derivatives z_t^1, \ldots, z_x^2 may be expressed through z_τ^1, \ldots, z_n^2 by the formulas

$$z_x^1 = z_\tau^1 x_\tau + z_n^1 t_\tau, \quad z_t^1 = z_\tau^1 t_\tau - z_n^1 x_\tau,$$
 (7)

etc. The system (6) then takes on the form

$$ho(z_{ au}^{1}t_{ au}-z_{n}^{1}x_{ au})=z_{ au}^{2}x_{ au}+z_{n}^{2}t_{ au},$$

$$k(z_{ au}^{1}x_{ au}+z_{n}^{1}t_{ au})=z_{ au}^{2}t_{ au}-z_{n}^{2}x_{ au},$$

or equivalently,

$$z_{n}^{1} = rac{
ho + k}{
ho x_{ au}^{2} - k t_{ au}^{2}} z_{ au}^{1} x_{ au} t_{ au} - rac{1}{
ho x_{ au}^{2} - k t_{ au}^{2}} z_{ au}^{2}, \ z_{n}^{2} = -rac{
ho k}{
ho x_{ au}^{2} - k t_{ au}^{2}} z_{ au}^{1} + rac{
ho + k}{
ho x_{ au}^{2} - k t_{ au}^{2}} z_{ au}^{2} x_{ au} t_{ au}.$$

Bearing in mind the continuity of z^1, z^2 , i.e., that of z_{τ}^1, z_{τ}^2 across the layers' interface, we average both equations (8) over the layers. The quantities $z_{\tau}^1, \ldots, z_n^2$ will then be replaced by $\langle z^1 \rangle_{\tau}, \ldots, \langle z^2 \rangle_n$ where the operation $\langle \cdot \rangle = m_1(\cdot)_1 + m_2(\cdot)_2$ is expressed through the volume fractions $m_1, m_2 \geq 0(m_1 + m_2 = 1)$ of materials 1 and 2 in the composite, and $(\cdot)_1, (\cdot)_2$ denote the values of (\cdot) in these materials.

Preserving the notation $z_{\tau}^1, \ldots, z_n^2$ for $\langle z^1 \rangle_{\tau}, \ldots, \langle z^2 \rangle_n$, we arrive at the following system:

$$z_n^1 = (A+B)z_{\tau}^1 x_{\tau} t_{\tau} - C z_{\tau}^2,$$

$$z_n^2 = -\mathcal{D} z_{\tau}^1 + (A+B)z_{\tau}^2 x_{\tau} t_{\tau},$$
(9)

with parameters A, \ldots, \mathcal{D} defined as

$$A = \langle \frac{k}{\rho x_{\tau}^{2} - kt_{\tau}^{2}} \rangle, \quad B = \langle \frac{\rho}{\rho x_{\tau}^{2} - kt_{\tau}^{2}} \rangle,$$

$$C = \langle \frac{1}{\rho x_{\tau}^{2} - kt_{\tau}^{2}} \rangle, \quad \mathcal{D} = \langle \frac{\rho k}{\rho x_{\tau}^{2} - kt_{\tau}^{2}} \rangle. \tag{10}$$

By virtue of Eqs. (7),(10), the system (9) may be reduced to

$$\alpha z_{x}^{1} t_{\tau} + \beta z_{t}^{1} x_{\tau} = z_{x}^{2} x_{\tau} + z_{t}^{2} t_{\tau},$$

$$z_{x}^{1} x_{\tau} + z_{t}^{1} t_{\tau} = \bar{\alpha} z_{x}^{2} t_{\tau} + \bar{\beta} z_{t}^{2} x_{\tau},$$
(11)

with

$$\alpha = A/C, \;\; \beta = B/C, \;\; \bar{\alpha} = A/\mathcal{D}, \; \bar{\beta} = B/\mathcal{D}.$$

Evidently,

$$\frac{\bar{\alpha}}{\alpha} = \frac{\bar{\beta}}{\beta} = \Theta = \frac{C}{\mathcal{D}},$$

and Eqs. (11) thus depend on three parameters: α, β, Θ . These parameters will be conveniently expressed through $\rho_i, k_i (i = 1, 2), m_1, m_2$, and V = dx/dt. To this end,

we first notice that

$$A = \langle rac{k}{
ho x_{ au}^2 - k t_{ au}^2}
angle = rac{m_1 k_1}{
ho_1 x_{ au}^2 - k_1 t_{ au}^2} + rac{m_2 k_2}{
ho_2 x_{ au}^2 - k_2 t_{ au}^2} = \ = rac{k_1 k_2}{(
ho_1 x_{ au}^2 - k_1 t_{ au}^2) (
ho_2 x_{ au}^2 - k_2 t_{ au}^2)} \left[(rac{ ilde{1}}{a^2}) x_{ au}^2 - t_{ au}^2
ight], \ B = \langle rac{
ho}{
ho x_{ au}^2 - k t_{ au}^2}
angle = rac{
ho_1
ho_2}{(
ho_1 x_{ au}^2 - k_1 t_{ au}^2) (
ho_2 x_{ au}^2 - k_2 t_{ au}^2)} (x_{ au}^2 - ilde{a}^2 t_{ au}^2), \ C = \langle rac{1}{
ho x_{ au}^2 - k t_{ au}^2}
angle = rac{1}{(
ho_1 x_{ au}^2 - k_1 t_{ au}^2) (
ho_2 x_{ au}^2 - k_2 t_{ au}^2)} (ilde{
ho} x_{ au}^2 - ilde{k} t_{ au}^2), \ \mathcal{D} = \langle rac{
ho k}{
ho x_{ au}^2 - k t_{ au}^2}
angle = rac{1}{(
ho_1 x_{ au}^2 - k_1 t_{ au}^2) (
ho_2 x_{ au}^2 - k_2 t_{ au}^2)} [
ho_1
ho_2 \langle k
angle x_{ au}^2 - k_1 k_2 \langle
ho
angle t_{ au}^2].$$

Here, $\tilde{a^2}=m_1a_2^2+m_2a_1^2, \langle k \rangle=m_1k_1+m_2k_2,$ etc. Parameters α,β,Θ will be expressed as

$$\alpha = \frac{A}{C} = k_1 k_2 \frac{(\tilde{1}_{a^2})V^2 - 1}{\tilde{\rho}V^2 - \tilde{k}},$$

$$\beta = \frac{B}{C} = \rho_1 \rho_2 \frac{V^2 - \tilde{a}^2}{\tilde{\rho}V^2 - \tilde{k}},$$

$$\Theta = \frac{C}{D} = \frac{\tilde{\rho}V^2 - \tilde{k}}{\rho_1 \rho_2 \langle k \rangle V^2 - k_1 k_2 \langle \rho \rangle}.$$
(12)

Consider the characteristic form

$$z_{xx}^{1}(x_{ au}^{2}-\Thetalpha^{2}t_{ au}^{2})+z_{tt}^{1}(t_{ au}^{2}-\Thetaeta^{2}x_{ au}^{2})+2z_{xt}^{1}(1-\Thetalphaeta)x_{ au}t_{ au}$$

associated with Eqs. (11). The characteristics of (11) have the slopes on the (x,t)-plane computed as the roots λ_1, λ_2 of

$$\lambda^2(t_\tau^2 - \Theta\beta^2 x_\tau^2) - 2\lambda(1 - \Theta\alpha\beta)x_\tau t_\tau + x_\tau^2 - \Theta\alpha^2 t_\tau^2 = 0,$$

or, with the notation

This equation has the roots

$$\lambda_1 = -rac{V - lpha\sqrt{\Theta}}{Veta\sqrt{\Theta} - 1}, \;\; \lambda_2 = rac{V + lpha\sqrt{\Theta}}{Veta\sqrt{\Theta} + 1}$$

with the product

$$\lambda_1\lambda_2=rac{V^2-\Thetalpha^2}{1-V^2\Thetaeta^2}.$$

The roots λ_1, λ_2 will be real if $\Theta > 0$. The latter condition will be satisfied as long as Ineq. (5) holds. If V = 0, then $\alpha = k_1 k_2 / \tilde{k}$, $\beta = \rho_1 \rho_2 \tilde{a}^2 / \tilde{k}$, $\Theta = \tilde{k} / k_1 k_2 \langle \rho \rangle$, and $\lambda_1 \lambda_2 = -\Theta \alpha^2 = -\langle \rho \rangle^{-1} \langle 1/k \rangle^{-1} < 0$. If $V = \infty$, then $\lambda_1 \lambda_2 = -1/\Theta \beta^2 = -\langle k \rangle \langle 1/\rho \rangle < 0$. We will investigate $\lambda_1 \lambda_2$ as the function of V^2 .

Referring to (12) we show that the following formulas hold:

$$V^2-\Thetalpha^2=rac{1}{\Delta}P,~~1-V^2\Thetaeta^2=rac{1}{\Delta}Q,~~1-\Thetalphaeta=rac{1}{\Delta}S$$

with

$$\begin{split} P &= P(V) = (\tilde{\rho}V^2 - \tilde{k}) \left[\left(\frac{\tilde{1}}{k} \right) V^2 - \left(\frac{\tilde{1}}{\rho} \right) \right] V^2 - a_1^2 a_2^2 \left[\left(\frac{\tilde{1}}{a^2} \right) V^2 - 1 \right]^2, \\ Q &= Q(V) = (\tilde{\rho}V^2 - \tilde{k}) \left[\left(\frac{\tilde{1}}{k} \right) V^2 - \left(\frac{\tilde{1}}{\rho} \right) \right] - \frac{V^2}{a_1^2 a_2^2} (V^2 - \tilde{a}^2)^2, \\ S &= S(V) = (\tilde{\rho}V^2 - \tilde{k}) \left[\left(\frac{\tilde{1}}{k} \right) V^2 - \left(\frac{\tilde{1}}{\rho} \right) \right] - (V^2 - \tilde{a}^2) \left[\left(\frac{\tilde{1}}{a^2} \right) V^2 - 1 \right], \\ \Delta &= \Delta(V) = (\tilde{\rho}V^2 - \tilde{k}) \left[\left(\frac{\tilde{1}}{k} \right) V^2 - \left(\frac{\tilde{1}}{\rho} \right) \right]. \end{split}$$

For P, Q, S we have alternative expressions

$$egin{aligned} P &= ilde{
ho} \left(rac{ ilde{1}}{k}
ight) \left[V^2 - rac{1}{ ilde{
ho} \left(rac{ ilde{i}}{k}
ight)}
ight] (V^2 - a_1^2) (V^2 - a_2^2), \ Q &= -rac{1}{a_1^2 a_2^2} \left[V^2 - ilde{k} \left(rac{ ilde{1}}{
ho}
ight)
ight] (V^2 - a_1^2) (V^2 - a_2^2), \ S &= \left[ilde{
ho} \left(rac{ ilde{1}}{k}
ight) - \left(rac{ ilde{1}}{a^2}
ight)
ight] (V^2 - a_1^2) (V^2 - a_2^2). \end{aligned}$$

Equation (13) now obtains the form

$$Q\lambda^2 2VS\lambda + P = 0,$$

or, equivalently,

$$-rac{1}{a_1^2a_2^2}\left[V^2- ilde{k}\left(rac{ ilde{1}}{
ho}
ight)
ight]\lambda^2-2V\left[ilde{
ho}\left(rac{ ilde{1}}{k}
ight)-\left(rac{ ilde{1}}{a^2}
ight)
ight]\lambda+ ilde{
ho}\left(rac{ ilde{1}}{k}
ight)\left[V^2-rac{1}{ ilde{
ho}\left(rac{ ilde{1}}{k}
ight)}
ight]=0.$$

The product $\lambda_1\lambda_2$ of its roots equals to

$$\lambda_1\lambda_2=- ilde
ho\left(rac{ ilde{1}}{k}
ight)a_1^2a_2^2rac{V^2-rac{1}{ ilde
ho\left(rac{ ilde{1}}{k}
ight)}}{V^2- ilde k\left(rac{ ilde{1}}{
ho}
ight)}.$$

By direct inspection we show that $(\Delta \rho = \rho_2 - \rho_1, etc.)$

$$ilde{k} ilde{
ho}\left(rac{ ilde{1}}{
ho}
ight)\left(rac{ ilde{1}}{k}
ight)=\left(1+rac{m_1m_2}{
ho_1
ho_2}(\Delta
ho)^2
ight)\left(1+rac{m_1m_2}{k_1k_2}(\Delta k)^2
ight)>1,$$

and, consequently,

$$\tilde{k}\left(\frac{\tilde{1}}{\rho}\right) > \frac{1}{\tilde{\rho}\left(\frac{\tilde{1}}{k}\right)}.$$
 (14)

If V^2 is so chosen that

$$\tilde{k}\left(\frac{\tilde{1}}{\rho}\right) > V^2 > \frac{1}{\tilde{\rho}\left(\frac{\tilde{1}}{k}\right)},\tag{15}$$

then $\lambda_1 \lambda_2 > 0$. The right Ineq. (15) will become consistent with (5) if we show that the inequality

$$\frac{1}{\tilde{\rho}\left(\frac{\tilde{1}}{k}\right)} < a_1^2 \tag{16}$$

is possible, and then choose V^2 satisfying

$$rac{1}{ ilde{
ho}\left(rac{ ilde{1}}{k}
ight)} < V^2 < a_1^2 < a_2^2.$$

Ineq. (16) is confirmed by direct calculation:

$$egin{split} L &= rac{1}{ ilde{
ho}\left(rac{ ilde{ ilde{1}}}{ ilde{k}}
ight)} - a_1^2 = rac{k_1k_2}{ ilde{
ho}\langle k
angle} - rac{k_1}{
ho_1} = rac{k_1}{
ho_1 ilde{
ho}\langle k
angle} [k_2
ho_1 - (m_1k_1 + m_2k_2)(m_1
ho_2 + m_2
ho_1)] = \ &= rac{m_1k_1}{
ho_1 ilde{
ho}\langle k
angle} [(1+m_2)k_2
ho_1 - (1-m_2)k_1
ho_2 - m_2(k_2
ho_2 + k_1
ho_1)] \ &= rac{m_1k_1}{
ho_1 ilde{
ho}\langle k
angle} [
ho_1\Delta k - k_1\Delta
ho - m_2\Delta k\Delta
ho]. \end{split}$$

This expression can be made negative even though we assume that $a_2^2 > a_1^2$, i.e.

$$a_2^2-a_1^2=rac{k_2}{
ho_1}-rac{k_1}{
ho_1}=rac{1}{
ho_1
ho_2}(
ho_1\Delta k-k_1\Delta
ho)>0.$$

To make L negative, it is necessary that $\Delta k \Delta \rho > \rho_1 \Delta k - k_1 \Delta \rho > 0$. For example, if $k_2 = 10, \rho_2 = 9, k_1 = \rho_1 = 1$, then $\rho_1 \Delta k - k_1 \Delta \rho - m_2 \Delta k \Delta \rho = 9 - 8 - 72m_2$, and this expression will be ≤ 0 if $m_2 \geq 1/72$. At the same time, the difference $k_2/\rho_2 - k_1/\rho_1$ should be positive, i.e. the ratio k/ρ should increase. Combined with $\Delta k \Delta \rho > 0$ this means that the increase may be due to that in k and less intensive increase (not decrease) in ρ , or due to the decrease in ρ and less intensive decrease (not increase) in k.

In observance of $L \leq 0$ and Ineq (15), the product $\lambda_1 \lambda_2 \geq 0$. We thus obtain laminates for which both effective phase velocities show the same sign. This sign can be changed to opposite if we switch from V to -V. Both velocities will be positive for r-laminates and negative for ℓ -laminates (Figs. 2a,2b).

<u>Remark 1</u>. Ineq. (16) will become impossible if we assume that $\rho_1 = \rho_2$. Then

$$\frac{1}{\rho_1\left(\frac{\tilde{1}}{k}\right)} - \frac{k_1}{\rho_1} = \frac{m_1}{\rho_1\left(\frac{\tilde{1}}{k}\right)} \left(1 - \frac{k_1}{k_2}\right) > 0$$

since $k_2 > k_1$. In this case, the product $\lambda_1 \lambda_2$ cannot be made positive. A similar result holds true when $k_1 = k_2$. We conclude that the roots of the same sign can be produced through a simultaneous variation of both density and stiffness.

Remark 2. Ineq. (16) cannot be replaced by a stronger inequality

$$ilde{k}\left(rac{ ilde{1}}{
ho}
ight) < a_1^2.$$

This one is impossible since (recall that $\Delta a^2 = a_2^2 - a_1^2 > 0$)

$$\begin{split} \tilde{k}\left(\frac{\tilde{1}}{\rho}\right) - a_1^2 &= \left(m_1k_2 + m_2k_1\right)\left(\frac{m_1}{\rho_2} + \frac{m_2}{\rho_1}\right) - \frac{k_1}{\rho_1} \\ &= m_1\left[\left(1 - m_2\right)\frac{k_2}{\rho_2} - \left(1 + m_2\right)\frac{k_1}{\rho_1} + m_2\left(\frac{k_2}{\rho_1} + \frac{k_1}{\rho_2}\right)\right] \\ &= m_1\left[\left(1 - m_2\right)a_2^2 - \left(1 + m_2\right)a_1^2 + m_2\left(a_2^2\frac{\rho_2}{\rho_1} + a_1^2\frac{\rho_1}{\rho_2}\right)\right] \\ &= \frac{m_1}{\rho_1\rho_2}\left\{\left(a_2^2 - a_1^2\right)\rho_1\rho_2 - m_2\left[\left(a_2^2 + a_1^2\right)\rho_1\rho_2 - a_2^2\rho_2^2 - a_1^2\rho_1^2\right]\right\} \\ &= \frac{m_1}{\rho_1\rho_2}\left\{\left(\Delta a^2\right)\rho_1\rho_2 - m_2\left[a_2^2\rho_2(\rho_1 - \rho_2) - a_1^2\rho_1(\rho_1 - \rho_2)\right]\right\} \\ &= \frac{m_1}{\rho_1\rho_2}\left\{\left(\Delta a^2\right)\rho_1\rho_2 + m_2\Delta\rho\left[a_1^2\rho_2 - a_1^2\rho_1 + \rho_2(\Delta a^2)\right]\right\} \\ &= \frac{m_1}{\rho_1\rho_2}\left\{\left(\Delta a^2\right)\rho_2\left[\rho_1 + m_2\Delta\rho\right] + m_2a_1^2(\Delta\rho)^2\right\} \\ &= \frac{m_1}{\rho_1\rho_2}\left\{\left(\Delta a^2\right)\rho_2\langle\rho\rangle + m_2a_1^2(\Delta\rho)^2\right\} > 0. \end{split}$$

Remark 3

The screening effect will be preserved if $\rho_2 = k_2$ and $\rho_1 = k_1$ but $\rho_1 \neq \rho_2$. Ineq. (5) will then be trivially satisfied by any value of V^2 since $a_1^2 = a_2^2 = 1$. By Ineq. (14),

$$\tilde{
ho}\left(\frac{\tilde{1}}{
ho}\right) > 1,$$

and Ineq. (15) specifies the interval of V^2 making the product $\lambda_1 \lambda_2$ positive. This case is special because it eliminates limitations on the admissible choice of V^2 .

6. The Analogy with Gas Dynamics

The effect observed here presents a close analogy with supersonic gas dynamics where disturbances travel with the speed a of sound relative to the particles and are at the same time transferred by the particles moving with their own material velocity v. Should |v| be greater than a, then the resulting speeds $v \pm a$ of disturbances relative to the laboratory system will be both of the same sign. In our case, a similar phenomenon emerges due to a local redistribution of mass density ρ as well as to the oscillations of the local stiffness k. These oscillations result in the effective decrease of phase velocity due to the successive reflections of waves from the layers' interfaces and the subsequent attenuation of the transient wave (Bakhvalov & Panasenko (1984)). Both effects work together toward the desired overall result allowing the disturbances to propagate in the same direction relative to the laboratory frame.

The collision of characteristics occurs both in gas dynamics and in a control problem. In gas dynamics, it creates the shock waves emerging in the flow in accordance with the non-linear equations of motion. Particularly, the relevant set of conservation laws (mass, momentum, energy) is observed across the shock waves.

In a control problem, the appearance of a strong discontinuity along the side mn of the rectangle mnpq on Fig. 3 cannot, however, be allowed from the viewpoint of the integrity of the construction. The damaging influence of this discontinuity can be avoided once the bar is originally cut at the location of this discontinuity. Both sides of the cut will then be exposed to their own independent displacements; the left half of the bar may be disregarded and taken away. There will be no reflected wave traveling back to the right from the cut: the energy of the initial state confined to the segment mq on Fig. 3 will be transformed into the work required to produce the relevant displacement of the left end of the bar over the time period ms.

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