

MULTI STEP ROCKET PROBLEM

due 9/25

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ROCKET PHASE

$$\theta = 59^\circ$$

$$t_r = 8.1 \text{ s}$$

$$a_r = 4.9 \text{ m/s}^2$$

$$v_{ri} = 0 \text{ m/s}$$

$$d_r = ?$$

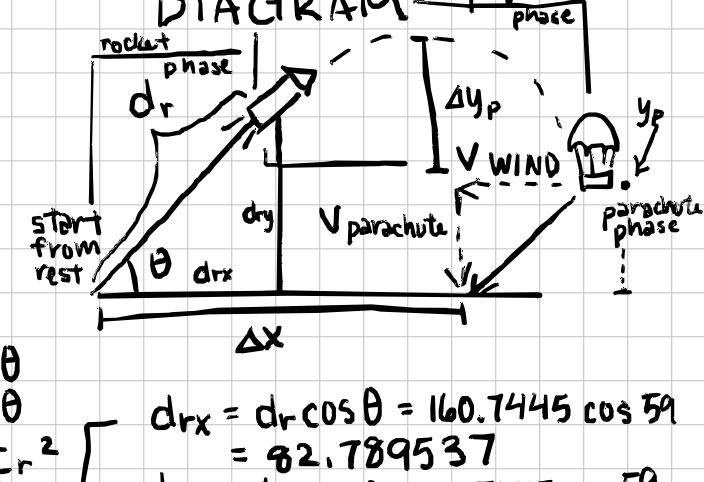
$$d_{rx} = d_r \cos \theta$$

$$d_{ry} = d_r \sin \theta$$

$$d_r = v_{ri} t_r + \frac{1}{2} a_r t_r^2$$

$$d_r = 0 + \frac{1}{2} \cdot 4.9 \cdot 8.1^2$$

$$d_r = 160.7445$$



$$V_{rf} = v_{ri} + a_r t_r \rightarrow V_{rf} = 36.69 \text{ m/s}$$

$$V_{rf} = 0 + 4.9 \cdot 8.1$$

- used no- V_f -equation (Δy version) to solve for distance traveled by accelerating rocket, then broke the distance into x and y components
- calculated final velocity of accelerating rocket to find starting velocity of projectile phase (using no- Δy -equation)

PROJECTILE PHASE

$$v_{pi} = V_{rf} = 36.69$$

$$\theta = 59^\circ$$

$$v_{px} = 36.69 \cos 59$$

$$\Delta x_p$$

$$t_p = \frac{\Delta x_p}{v_{px}}$$

$$v_{pyi} = 36.69 \sin 59 \text{ m/s}$$

$$a_p = -9.8 \text{ m/s}^2$$

$$t_p$$

$$\Delta y_{p0} = ?$$

$$\Delta y_{pt} = -78 \text{ m}$$

to reach max height,

$$v_{pyf} = 0 \text{ m/s}$$

$$v_{pyf}^2 = v_{pyi}^2 + 2a_p \Delta y_p$$

$$0 = (36.69 \sin 59)^2 - 19.6 \Delta y_{p1}$$

$$\Delta y_{p1} = -50.462707$$

$$\Delta y_{pt} - \Delta y_{p1} = \Delta y_{p0} \leftarrow -78 + 50.462707 = \Delta y_{p0}$$

$$\Delta y_{p0} = -27.537293 \text{ m}$$

- used no-t-equation to

find the Δy from the start of

the projectile to the max height (so $v_{pyf} = 0 \text{ m/s}$)

- the overall Δy of the projectile phase only takes into account the change from the starting position, not the Δy from starting h to max h, so $\Delta y_{p0} = \Delta y_{pt} - \Delta y_{p1}$ (refer to projectile phase diagram)

$$\Delta y = v_i t + \frac{1}{2} a t^2$$

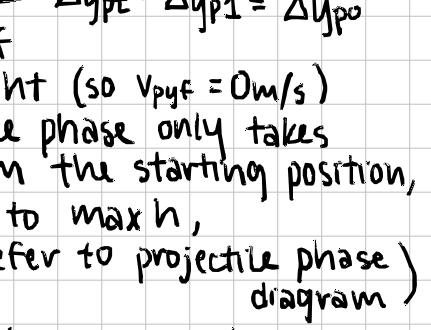
$$\Delta y_{p0} = v_{pi} \cdot \frac{\Delta x_p}{v_{px}} + \frac{a}{2} \left(\frac{\Delta x_p}{v_{px}} \right)^2$$

$$-27.537293 = \frac{\sin 59}{\cos 59} \cdot \Delta x_p - 4.9 \frac{\Delta x_p^2}{(36.69 \cos 59)^2}$$

$$0 = -0.013721444 \Delta x_p^2 + 1.664279482 \Delta x_p + 27.537293$$

$$\Delta x_p = 136.03602 \text{ m}$$

PROJECTILE PHASE DIAGRAM



- used no- V_f -equation (Δy version) to solve for Δx of the overall projectile, so Δy_{p0} was used as the

$$\Delta y$$

PARACHUTE PHASE

- the current y position is the sum of the overall Δy 's of the past two phases (rocket) (projectile)

$$d_{ry} + \Delta y_{p0} = y_{\text{parachute initial}}$$

$$137.78492 - 27.537293 = y_{ci}$$

$$y_{pi} = 110.24763 \text{ m}$$

- in order to get back to the ground ($y_f = 0$), the $\Delta y_{\text{parachute}}$ is -110.24763 m

$$\frac{x}{\Delta x_c} = \frac{y}{\Delta y_c} = \frac{\Delta x_c}{\Delta y_c} = \frac{v_{cx}}{v_{cy}}$$

$$v_{cx} = -17 \text{ m/s} \quad v_{cy} = -7 \text{ m/s}$$

$$t_c = \frac{\Delta x_c}{v_{cx}} \quad t_c =$$

$$\Delta y_c = v_{cy} t_c$$

$$\Delta y_c = v_{cy} \frac{\Delta x_c}{v_{cx}}$$

$$-110.24763 = \frac{-7}{-17} \Delta x_c$$

$$\Delta x_c = -267.74424 \text{ m}$$

↓

48.92 m West

- used equation for constant v and substituted $\frac{\Delta x_c}{v_{cx}}$ for t_c to solve for Δx_c using one equation

for <math